

A short course on

# Nonlinear Finite Element Analysis

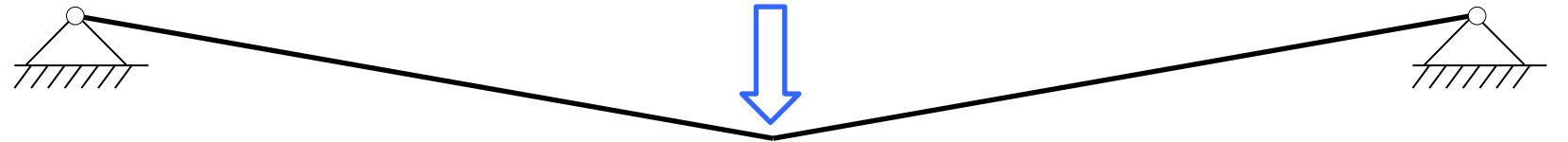
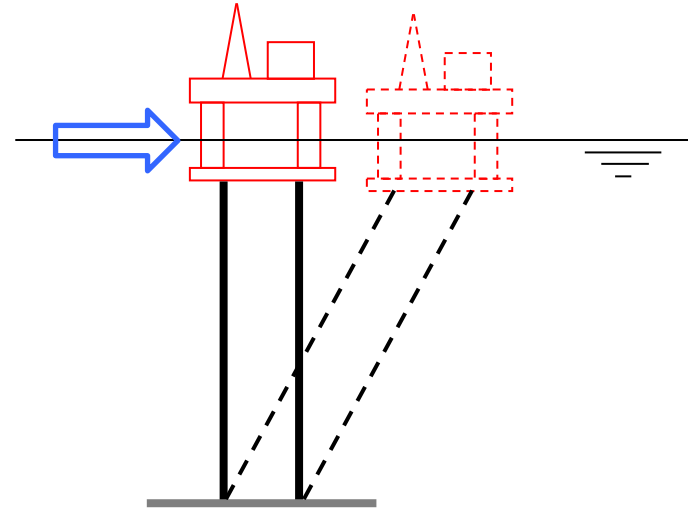
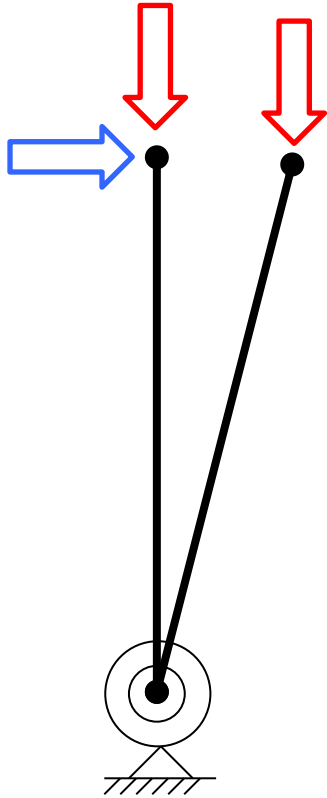
This video:

**Geometric Nonlinearity**

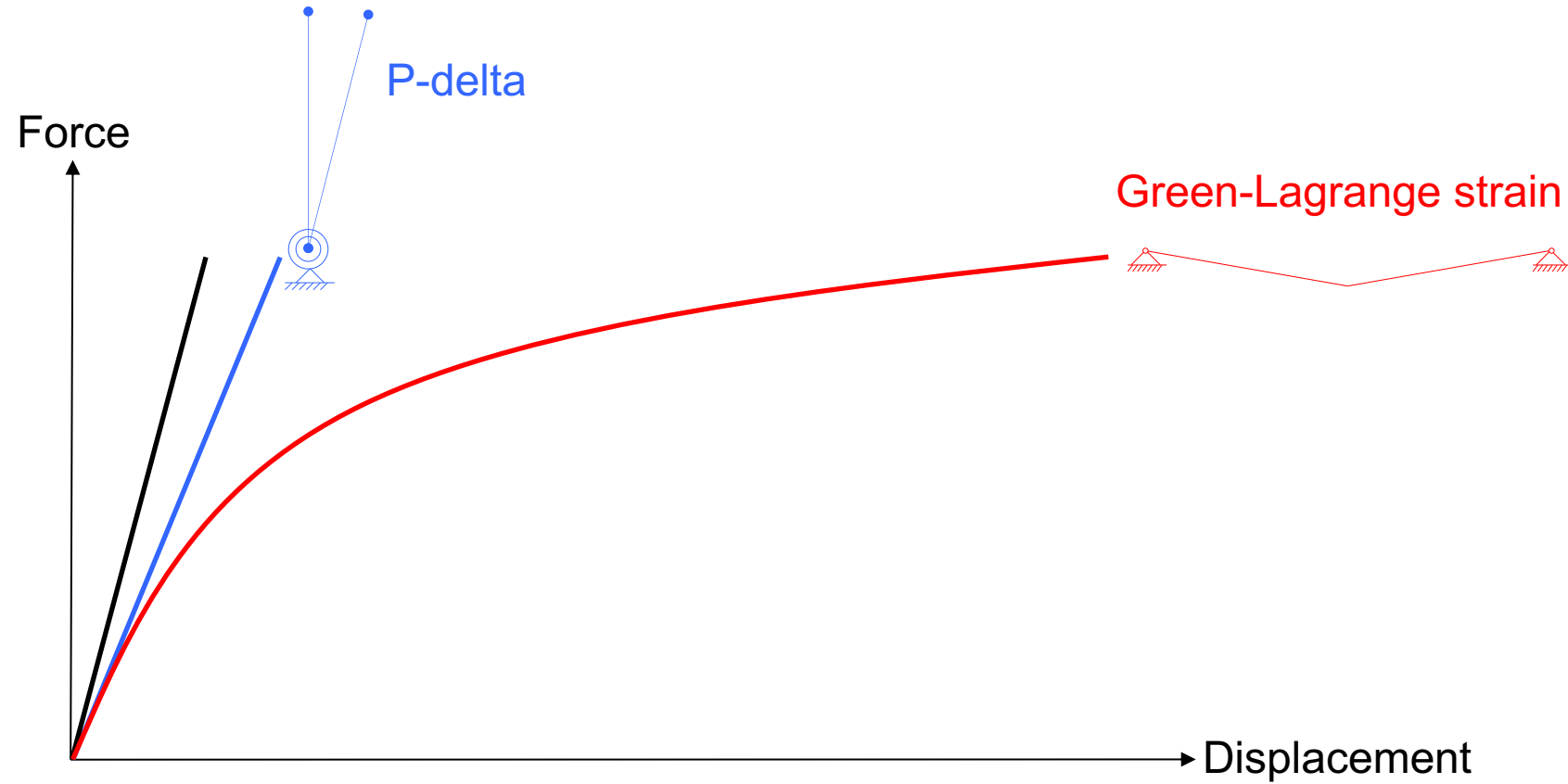
Terje's Toolbox is freely available at [terje.civil.ubc.ca](http://terje.civil.ubc.ca)

It is created and maintained by Professor Terje Haukaas, Ph.D., P.Eng.,  
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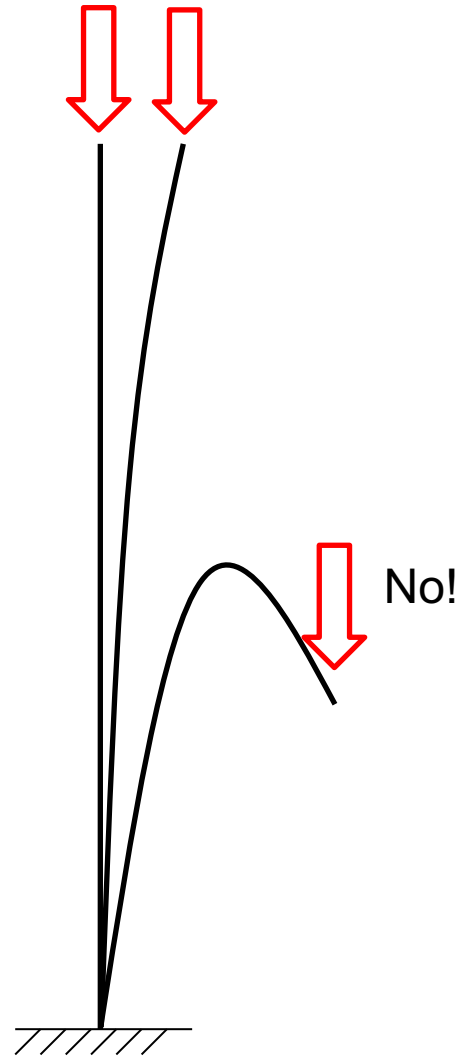
# Effects of Changing Geometry



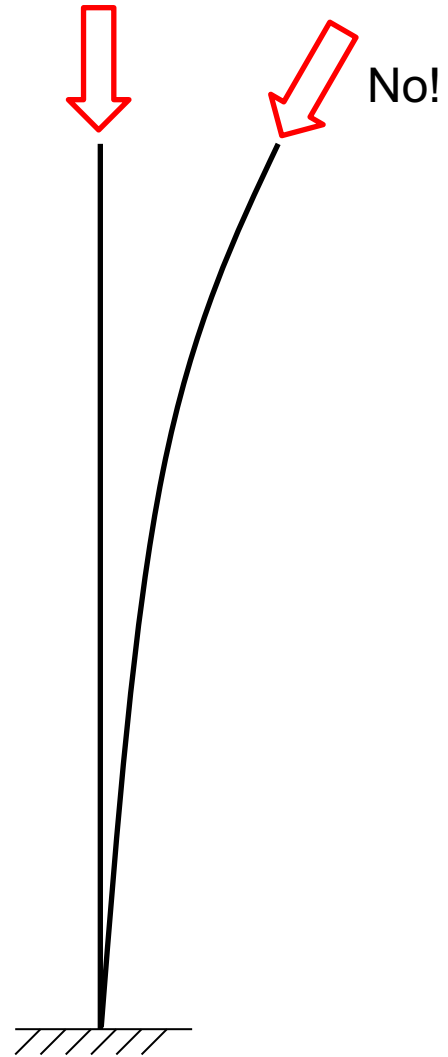
# Linear or Nonlinear



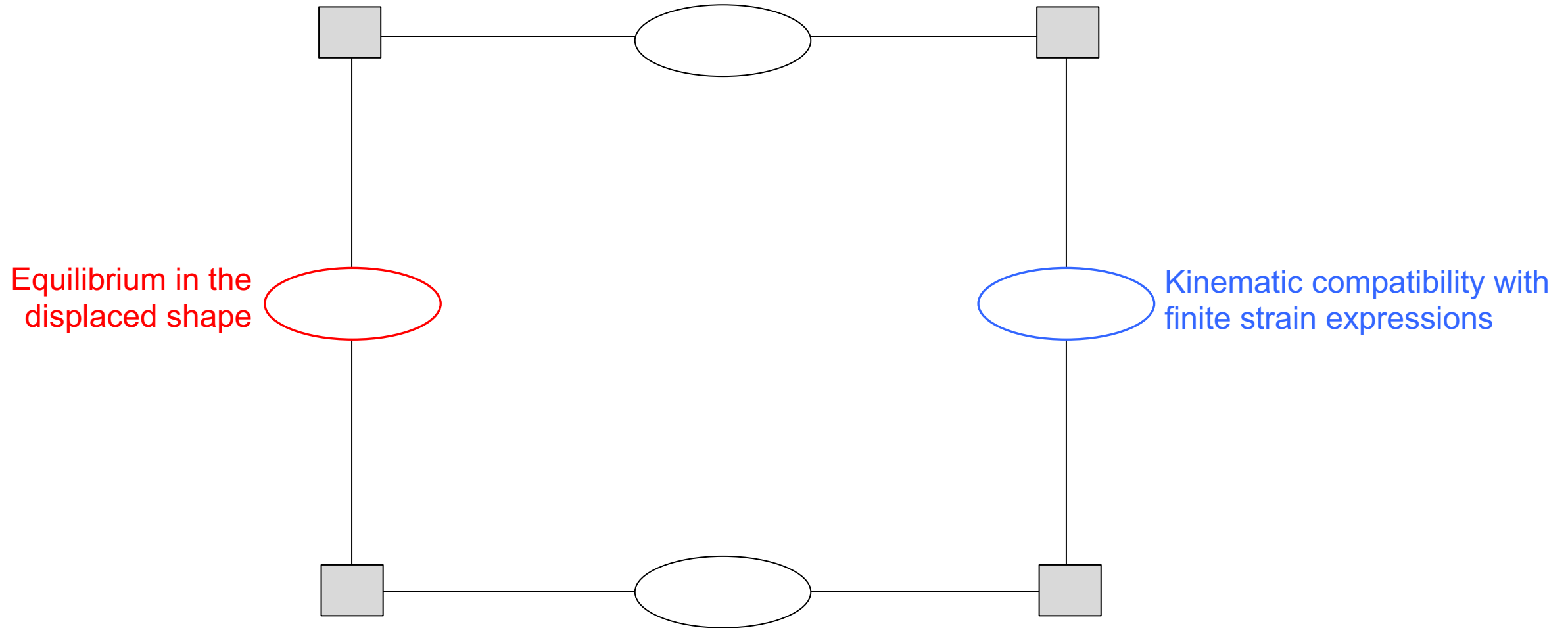
# Only Moderate Deformations



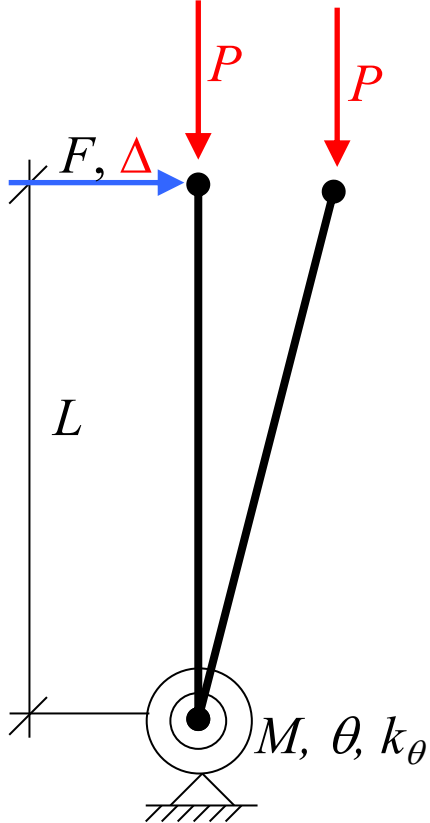
# Only Conservative Loads



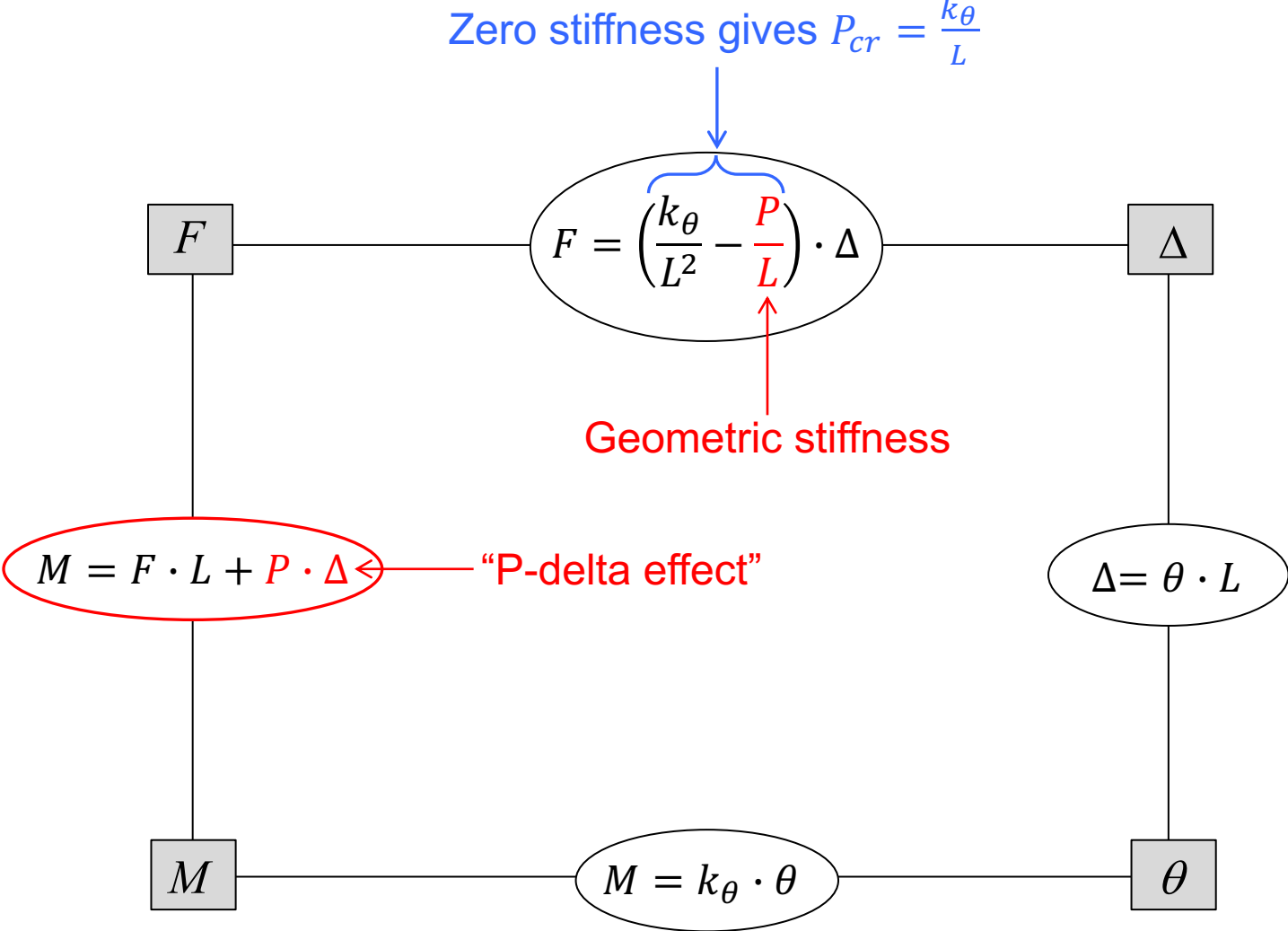
# How to Include



# Leaning Column

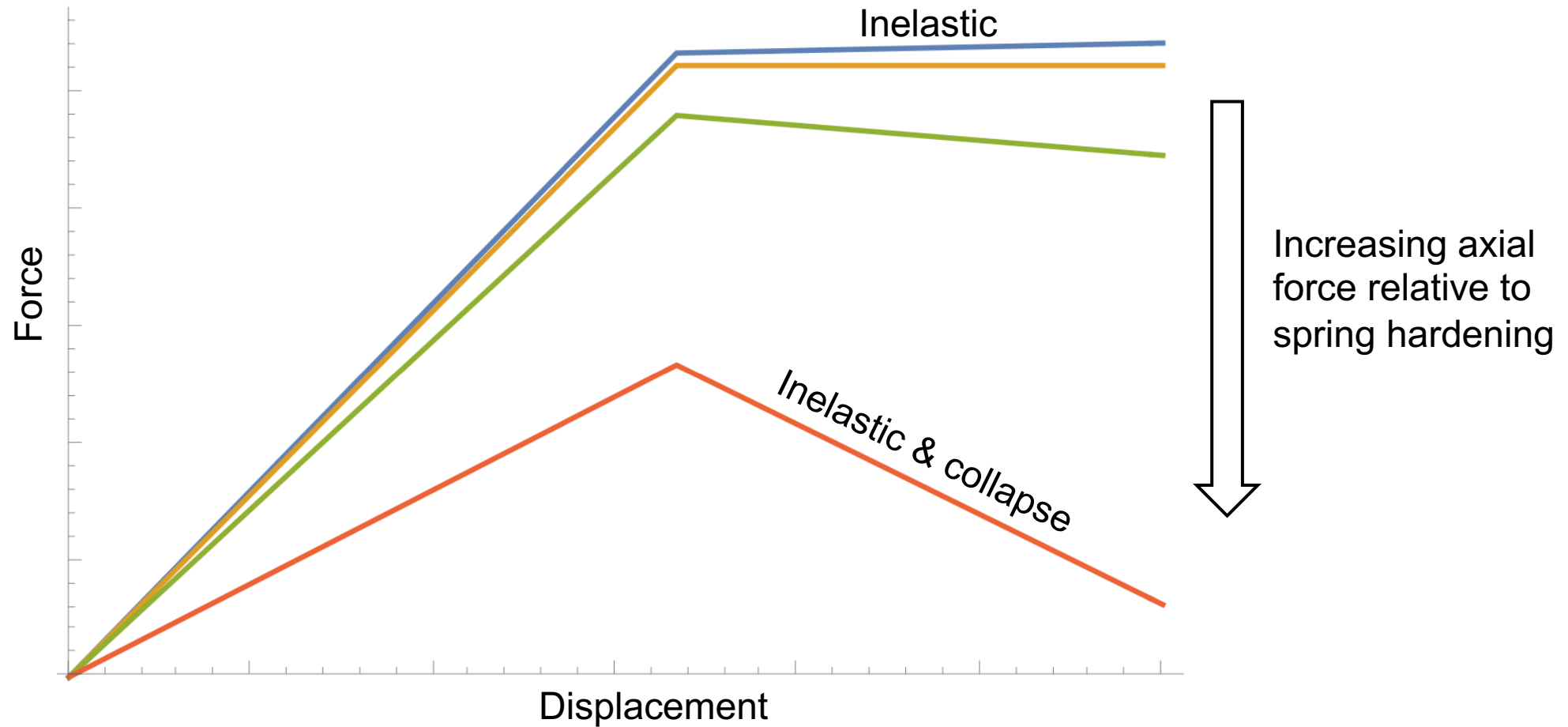


Zero stiffness gives  $P_{cr} = \frac{k_\theta}{L}$



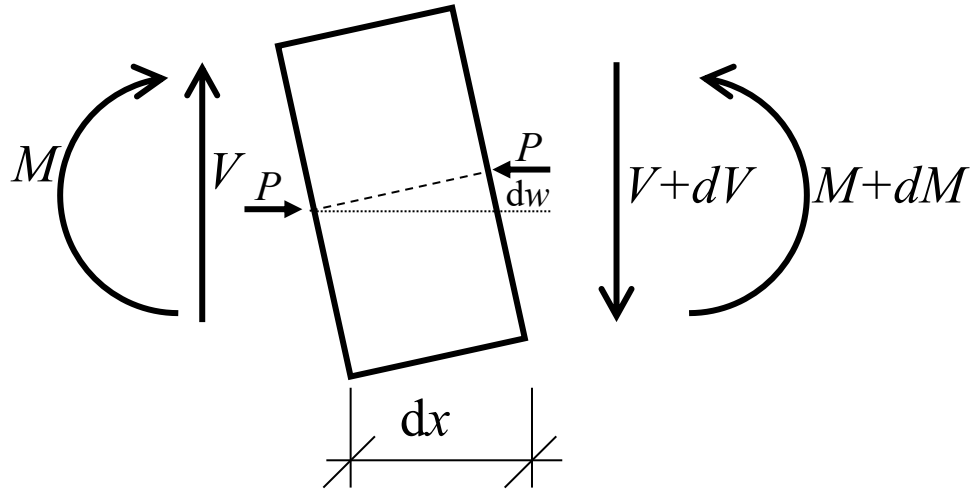
# Stick Model with P-Delta

See example at [terje.civil.ubc.ca](http://terje.civil.ubc.ca)





# Beam with Axial Force



Equilibrium of displaced element:

$$V = \frac{dM}{dx} + P \cdot \frac{dw}{dx}$$

Revised differential equation:

$$EI \cdot \frac{d^4 w}{dx^4} + P \cdot \frac{d^2 w}{dx^2} = q_z$$

# Element 6

$$\int_0^L (EI \cdot w'''' + P \cdot w'') \delta w dx = 0$$

$$\int_0^L EI \cdot w'' \delta w'' dx - \int_0^L P \cdot w' \delta w' dx = 0$$

$$\left( \int_0^L EI \cdot \mathbf{N}''^T \cdot \mathbf{N}'' dx - P \cdot \underbrace{\int_0^L \mathbf{N}'^T \cdot \mathbf{N}' dx}_{\mathbf{k}_G} \right) \cdot \mathbf{u} = 0$$

$$\mathbf{K} = \begin{bmatrix} \frac{12EI}{L^3} & -\frac{6EI}{L^2} & \frac{12EI}{L^3} & -\frac{6EI}{L^2} \\ -\frac{6EI}{L^2} & \frac{4EI}{L} & \frac{6EI}{L^2} & \frac{2EI}{L} \\ -\frac{12EI}{L^3} & \frac{6EI}{L^2} & \frac{12EI}{L^3} & \frac{6EI}{L^2} \\ -\frac{6EI}{L^2} & \frac{2EI}{L} & \frac{6EI}{L^2} & \frac{4EI}{L} \end{bmatrix} - P \cdot \underbrace{\begin{bmatrix} \frac{6}{5L} & -\frac{1}{10} & -\frac{6}{5L} & -\frac{1}{10} \\ -\frac{1}{10} & \frac{2L}{15} & \frac{1}{10} & -\frac{L}{30} \\ -\frac{6}{5L} & \frac{1}{10} & \frac{6}{5L} & \frac{1}{10} \\ -\frac{1}{10} & -\frac{L}{30} & \frac{1}{10} & \frac{2L}{15} \end{bmatrix}}_{\mathbf{K}_G}$$



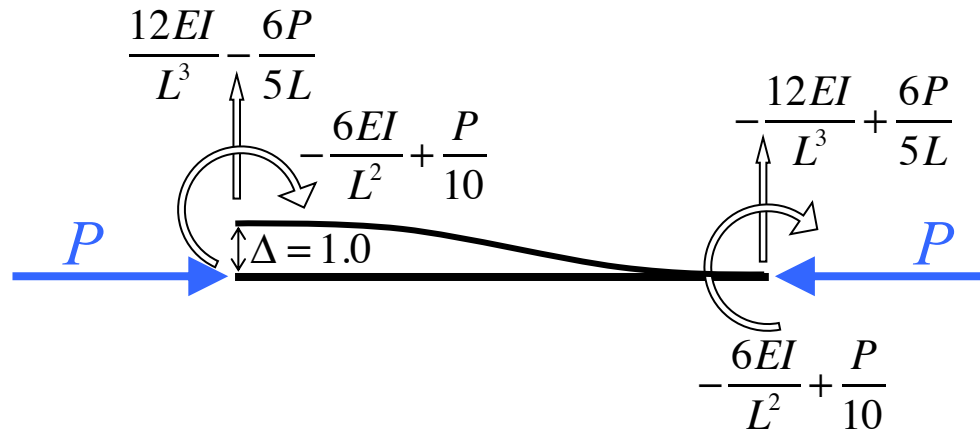
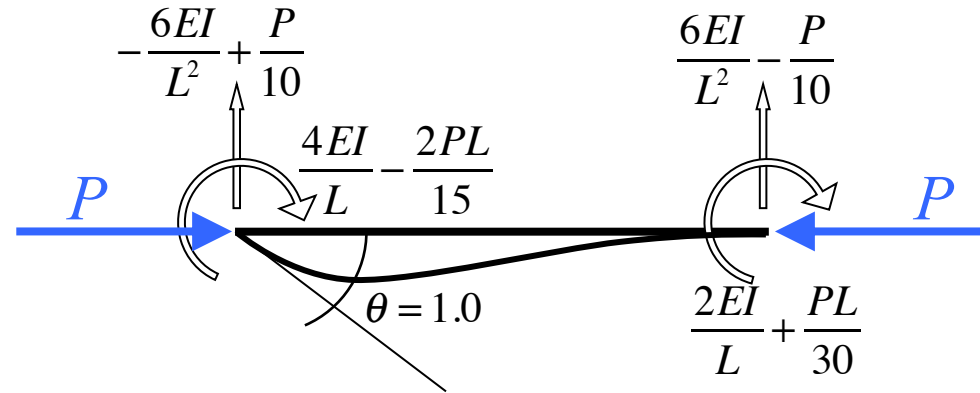
$$N_1(x) = \frac{2x^3}{L^3} - \frac{3x^2}{L^2} + 1$$

$$N_2(x) = -\frac{x^3}{L^2} + \frac{2x^2}{L} - x$$

$$N_3(x) = -\frac{2x^3}{L^3} + \frac{3x^2}{L^2}$$

$$N_4(x) = -\frac{x^3}{L^2} + \frac{x^2}{L}$$

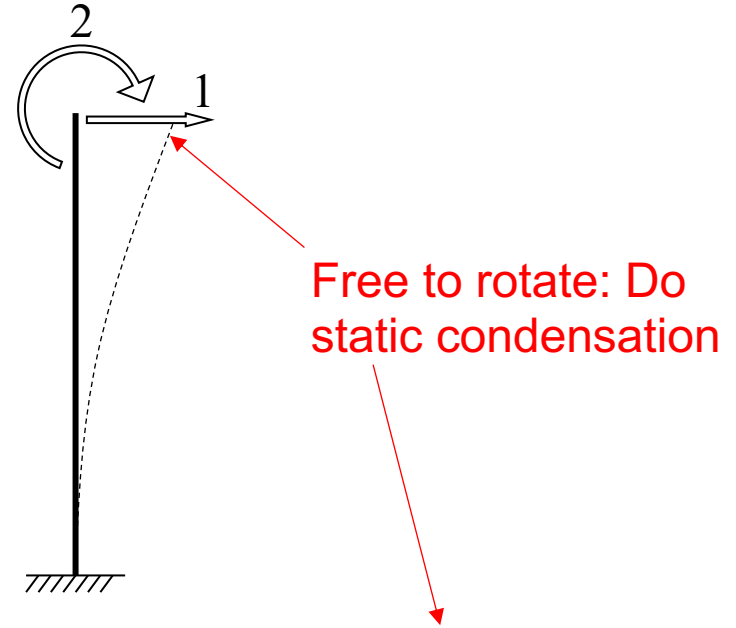
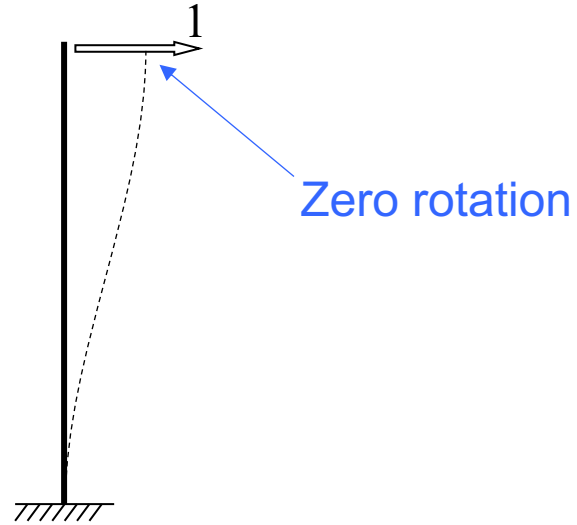
# Alternate View



# Big & Small P-Delta

$$\mathbf{K}_G = P \cdot \begin{bmatrix} \frac{1}{5L} & -\frac{1}{10} & -\frac{1}{5L} & -\frac{1}{10} \\ -\frac{1}{10} & \frac{2L}{15} & \frac{1}{10} & -\frac{L}{30} \\ -\frac{1}{5L} & \frac{1}{10} & \frac{1}{5L} & \frac{1}{10} \\ -\frac{1}{10} & -\frac{L}{30} & \frac{1}{10} & \frac{2L}{15} \end{bmatrix} + \begin{bmatrix} \frac{P}{L} & 0 & -\frac{P}{L} & 0 \\ 0 & 0 & 0 & 0 \\ -\frac{P}{L} & 0 & \frac{P}{L} & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}$$

# Cantilever



Elastic:

$$K_o = \frac{12EI}{L^3}$$

$$K_o = \frac{12EI}{L^3} - \left(-\frac{6EI}{L^2}\right) \left(\frac{L}{4EI}\right) \left(-\frac{6EI}{L^2}\right) = \frac{3EI}{L^3}$$

Big P-delta:

$$K_G = -\frac{P}{L}$$

$$K_G = -\frac{P}{L}$$

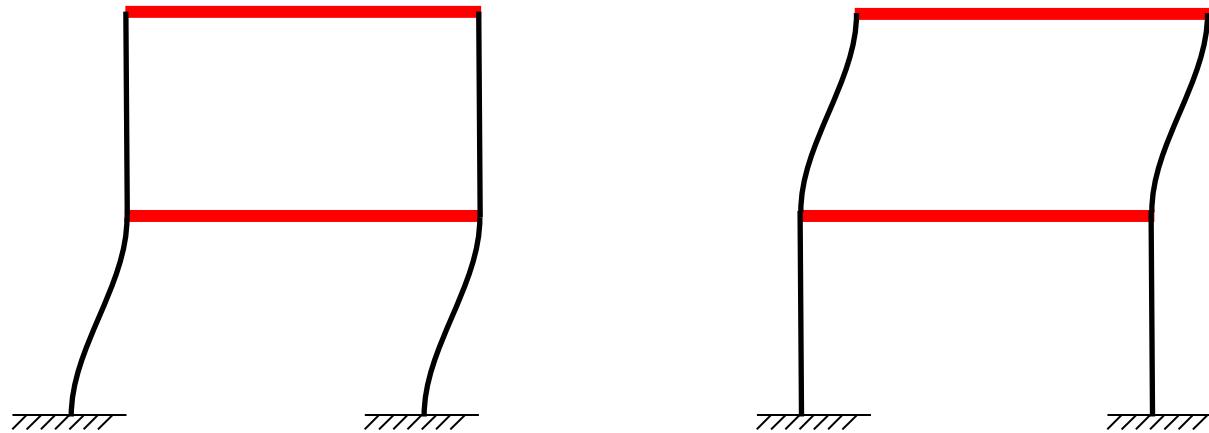
Small and big P-delta:

$$K_G = -\frac{6P}{5L}$$

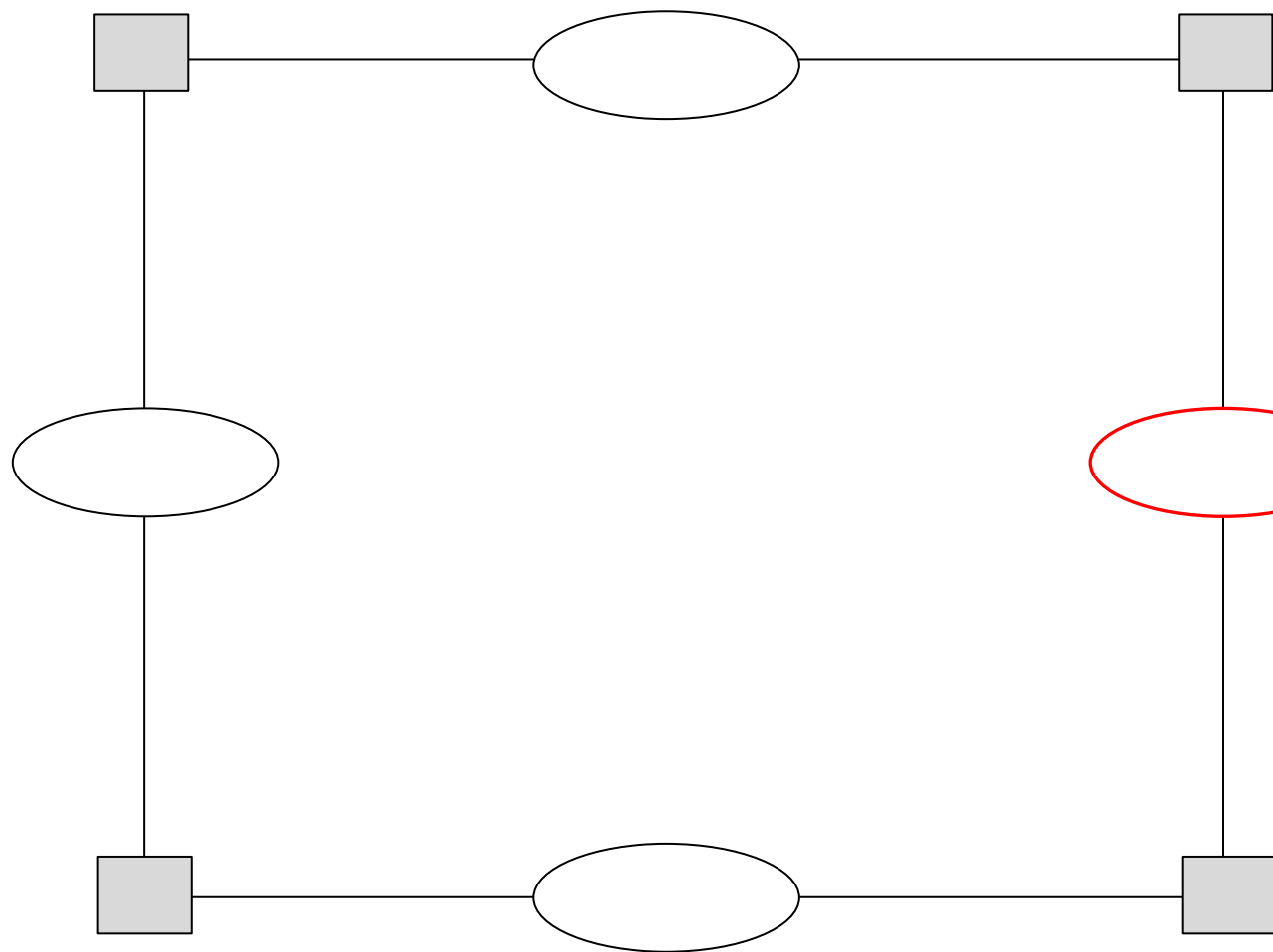
$$K_G = \frac{6P}{5L} - \left(\frac{P}{10}\right) \left(-\frac{15}{2PL}\right) \left(\frac{P}{10}\right) = -\frac{9P}{8L}$$

# Buckling Loads & Modes

Generalized eigenvalue problem:  $[\mathbf{K} - P \cdot \mathbf{K}_G] \mathbf{u} = \mathbf{F}$



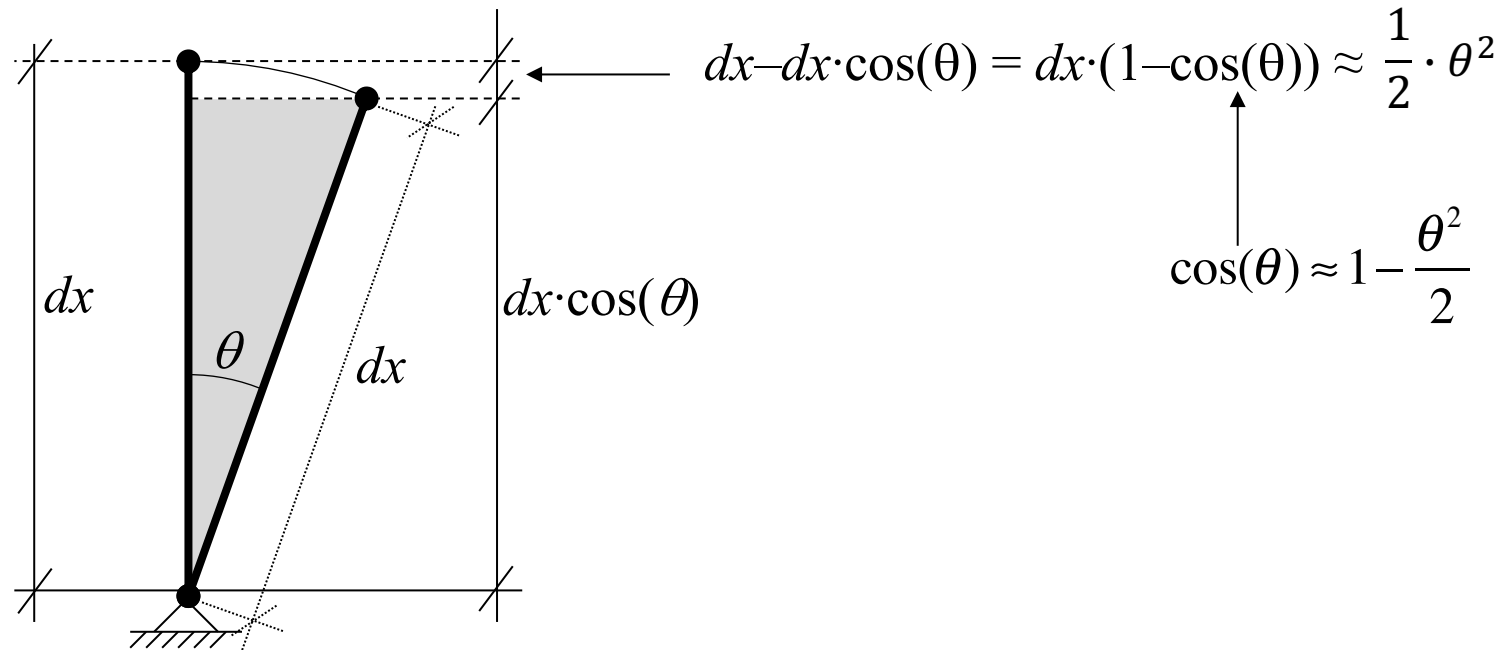
# Revised Compatibility



Finite strain expression  
+ Principle of virtual displacements  
= Finite elements with geometric nonlinearity

# Green-Lagrange Strain

$$E_{xx} = \frac{1}{2} (2 \cdot u_{x,x} + u_{x,x}u_{x,x} + u_{z,x}u_{z,x}) = \frac{du}{dx} + \frac{1}{2} \left( \frac{du}{dx} \right)^2 + \frac{1}{2} \left( \frac{dw}{dx} \right)^2$$





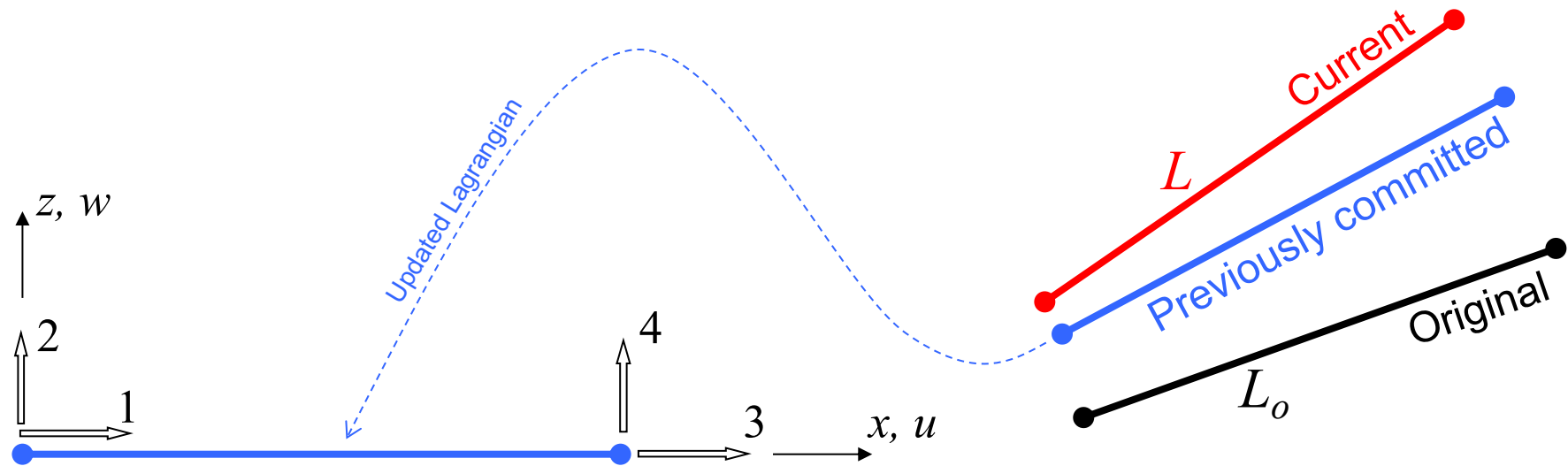
# Element 3

Green's strain:

$$\varepsilon = \frac{1}{2} \left( \left( \frac{L}{L_0} \right)^2 - 1 \right) \leftrightarrow \varepsilon = \frac{du}{dx} + \frac{1}{2} \cdot \left( \frac{dw}{dx} \right)^2$$

Coordinate system:

Total Lagrangian **or** Updated Lagrangian **or** Corotational



# Basic to Local

Principle of virtual displacements:  $\int_V \sigma \delta \varepsilon dV = \int_0^L \mathbf{p}^T \delta \tilde{\mathbf{u}} dx$

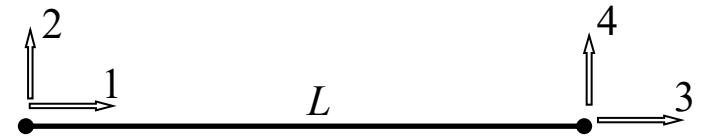
Material law & section integration ( $\sigma = E\varepsilon$ ):  $EA \cdot \int_0^L \varepsilon \delta \varepsilon dx = \int_0^L \mathbf{p}^T \delta \tilde{\mathbf{u}} dx$

$$\varepsilon = u' + \frac{1}{2} \cdot w'^2$$

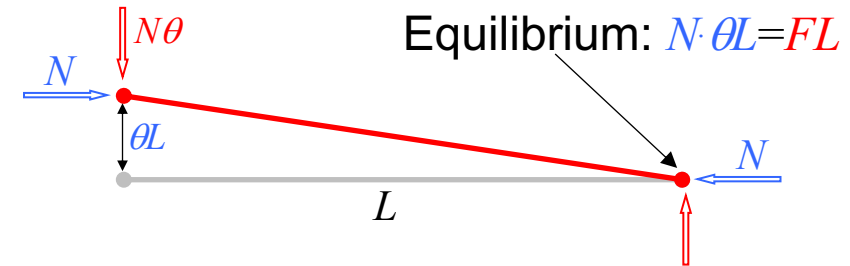
$$\begin{cases} u(x) \\ w(x) \end{cases} = \begin{bmatrix} N_1 & 0 & N_2 & 0 \\ 0 & N_1 & 0 & N_2 \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \\ u_4 \end{Bmatrix}$$

$N_1(x) = 1 - \frac{x}{L}$  and  $N_2(x) = \frac{x}{L}$

Variational calculus on Green's strain:  $\delta \varepsilon = \delta \left( u' + \frac{1}{2} \cdot w'^2 \right) = \delta u' + w' \cdot \delta w'$



Result:  $\tilde{\mathbf{F}} = \begin{Bmatrix} -1 \\ 0 \\ 1 \\ 0 \end{Bmatrix} N + \begin{bmatrix} 0 & 0 & 0 & 0 \\ 0 & \frac{1}{L} & 0 & -\frac{1}{L} \\ 0 & 0 & 0 & 0 \\ 0 & -\frac{1}{L} & 0 & \frac{1}{L} \end{bmatrix} \begin{Bmatrix} u_1 \\ u_2 \\ u_3 \\ u_4 \end{Bmatrix} N = \begin{Bmatrix} -1 \\ 0 \\ 1 \\ 0 \end{Bmatrix} N + \begin{Bmatrix} 0 \\ -\frac{(u_4 - u_2)}{L} \\ 0 \\ \frac{(u_4 - u_2)}{L} \end{Bmatrix} N = \begin{Bmatrix} -1 \\ -\theta \\ 1 \\ \theta \end{Bmatrix} N$



Equilibrium in the displaced shape and also  $\mathbf{T}_{b1}^T$

# Code

```
# Original element length
dx0 = xyz[1,:] - xyz[0,:]
L0 = np.sqrt(dx0.dot(dx0))

# Current element length
du = U[[2, 3], 0] - U[[0, 1], 0]
dx = dx0 + du
L = np.sqrt(dx.dot(dx))

# Green-Lagrange strain
es = 0.5 * (L**2 / L0**2 - 1)

# Force and stiffness in the basic system
kb = self.E * self.A / L
self.sTrial = self.E * self.A * es + self.NO

# Previously committed geometry
duStep = U[[2, 3], 1] - U[[0, 1], 1]
dxCommitted = dx0 + du - duStep
LCommitted = np.sqrt(dxCommitted.dot(dxCommitted))

# Direction cosines for transformation matrices
dxCommitted = dxCommitted / LCommitted
duStep = duStep / L

# Rotation of member since last commit (cross-product of unit vectors is sin(theta)~theta for small rotations)
theta = (duStep).dot([-dxCommitted[1], dxCommitted[0]])

# Transformation from Basic to Local, with Local being last commit
Tb1 = np.array([-1, -theta, 1, theta])

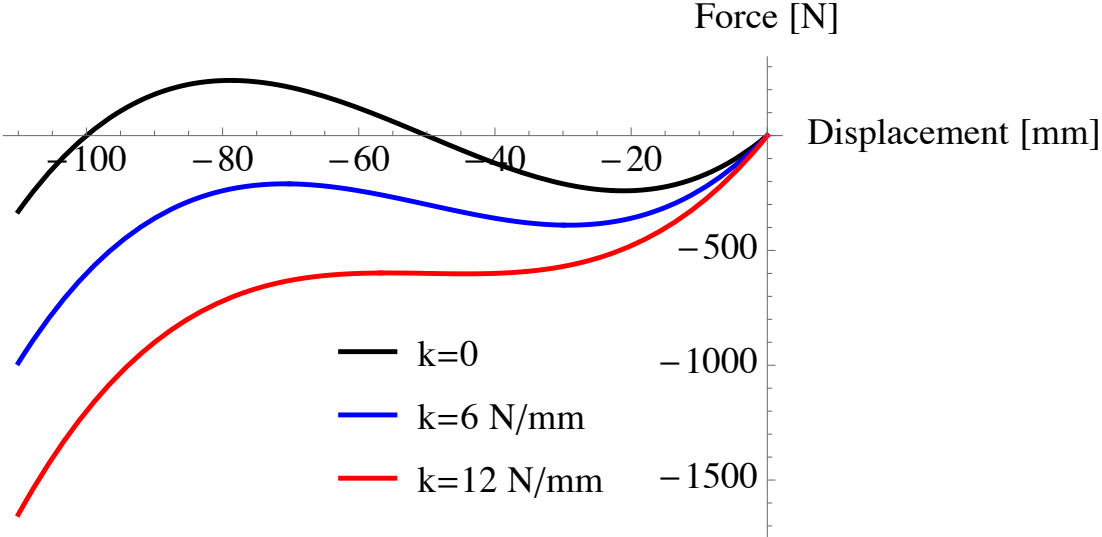
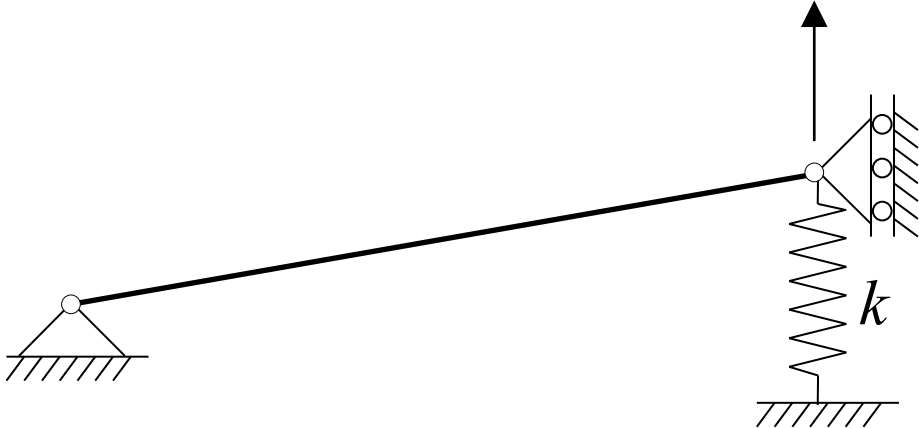
# Elastic element stiffness and force in Local configuration
Kl = np.outer(Tb1.dot(kb), Tb1)
tildeF1 = Tb1.dot(self.sTrial)

# Add geometric stiffness matrix in the Local configuration
c = np.array([0, -1, 0, 1])
kGeometric = np.outer(c, c) * self.sTrial / L
Kl += kGeometric

# Transform from Local to Global
Tlg = np.array([[dxCommitted[0], dxCommitted[1], 0.0, 0.0],
               [-dxCommitted[1], dxCommitted[0], 0.0, 0.0],
               [0.0, 0.0, dxCommitted[0], dxCommitted[1]],
               [0.0, 0.0, -dxCommitted[1], dxCommitted[0]]])
tildeFg = (np.transpose(Tlg)).dot(tildeF1)
Kg = ((np.transpose(Tlg)).dot(Kl)).dot(Tlg)
```

# Snap Through

See analytical example and also a G2 example at [terje.civil.ubc.ca](http://terje.civil.ubc.ca)



# Forms of the BVP

$$EI \cdot w'''' - q = 0$$

Weight and integrate

Require point-wise fulfilment

$$\int_0^L (EI \cdot w'''' - q) \cdot \delta w \cdot dx = 0$$

Integration by parts

Integration by parts

$$\int_0^L EI \cdot w'' \delta w'' dx = \int_0^L q \cdot \delta w dx$$

Anti-variation

Variation

$$\int_0^L \frac{1}{2} \cdot EI \cdot (w'')^2 dx = \int_0^L q \cdot w dx$$

# Energy Principles

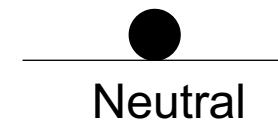
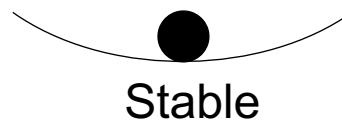
Total energy = Strain energy + Potential energy in applied loads

$$\Pi = U + H$$

Functionals:  $\Pi(w(x)) = U(w(x)) - H(w(x))$

Principle of minimum potential energy:  $\delta\Pi = 0$

Stability:  $\delta^2\Pi > 0$



# Ritz

Guess  $w(x)$  that satisfies the “essential” kinematic boundary conditions

$$w(x) = a \cdot \left(1 - \cos\left(\frac{n \cdot \pi \cdot x}{2L}\right)\right)$$

$$w(x) = a \cdot \left(\sin\left(\frac{n \cdot \pi \cdot x}{2L}\right)\right)$$

$$w(x) = \sum a_n \cdot \left(1 - \cos\left(\frac{n \cdot \pi \cdot x}{2L}\right)\right)$$

Principle of minimum potential energy + calculus of variation:

$$\delta\Pi = \frac{\partial\Pi}{\partial a} \cdot \delta a = 0$$

Must be zero!

Arbitrary variations!

Solve for  $a$ , or solve for the buckling load if  $a$  cancels

# Energy Expressions

- In general: 
$$U = \int_V \frac{1}{2} \cdot E \cdot (\varepsilon)^2 dV$$

$$\varepsilon = z \cdot \frac{d^2w}{dx^2} + \frac{1}{2} \cdot \left(\frac{dw}{dx}\right)^2$$

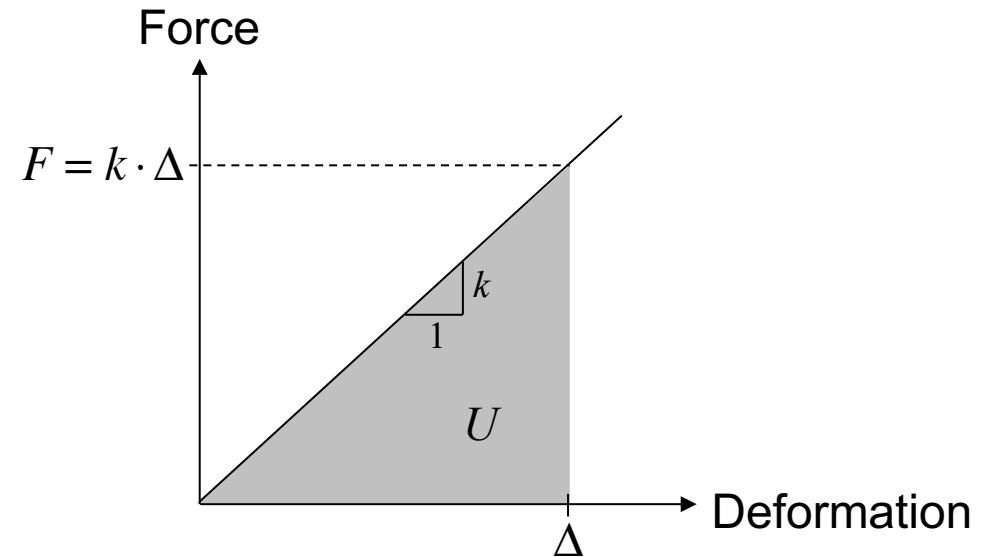
- Elastic spring: 
$$U = \int_0^{\Delta} F d\Delta = \int_0^{\Delta} (k \cdot \Delta) d\Delta = \frac{1}{2} \cdot k \cdot \Delta^2$$

- Beam bending: 
$$U = \int_0^{w''} EI \cdot w'' dw'' = \frac{1}{2} \cdot EI \cdot (w'')^2$$

- Point load: 
$$H = -F \Delta$$

- Distributed load: 
$$H = -\int_0^L q_z \cdot w dx$$

- Axial force: 
$$H = -\int_0^L P du = -P \cdot \int_0^L \frac{1}{2} \cdot (w')^2 dx$$





# Buckling of Simply Supported Column

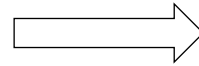
See example at [terje.civil.ubc.ca](http://terje.civil.ubc.ca)

$$U = \int_0^{w''} EI \cdot w'' dw'' = \frac{1}{2} \cdot EI \cdot (w'')^2$$

$$H = -\int_0^L P du = -P \cdot \int_0^L \frac{1}{2} \cdot (w')^2 dx$$

$$w(x) = a \cdot \sin\left(\frac{\pi}{L} \cdot x\right)$$

$$\delta\Pi = 0$$

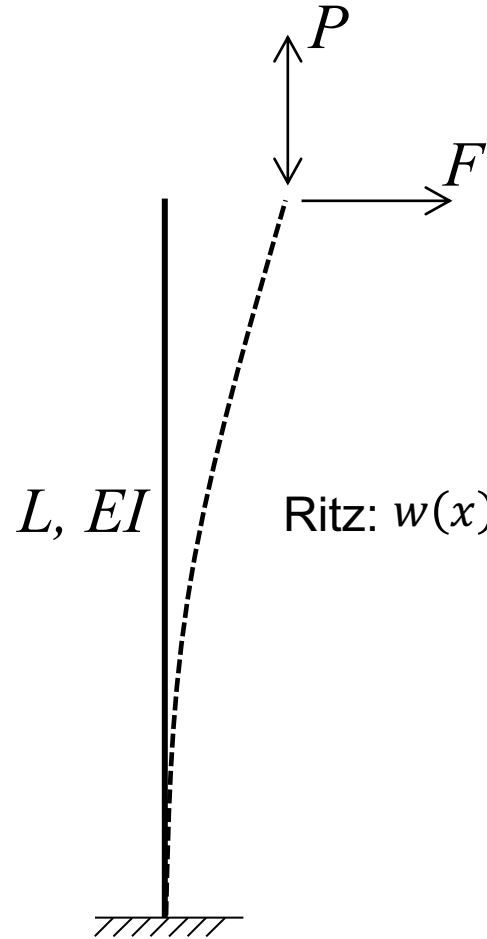


$$P_{cr} = \frac{\pi^2 \cdot EI}{L^2}$$

Exact Euler load because  
of good Ritz guess

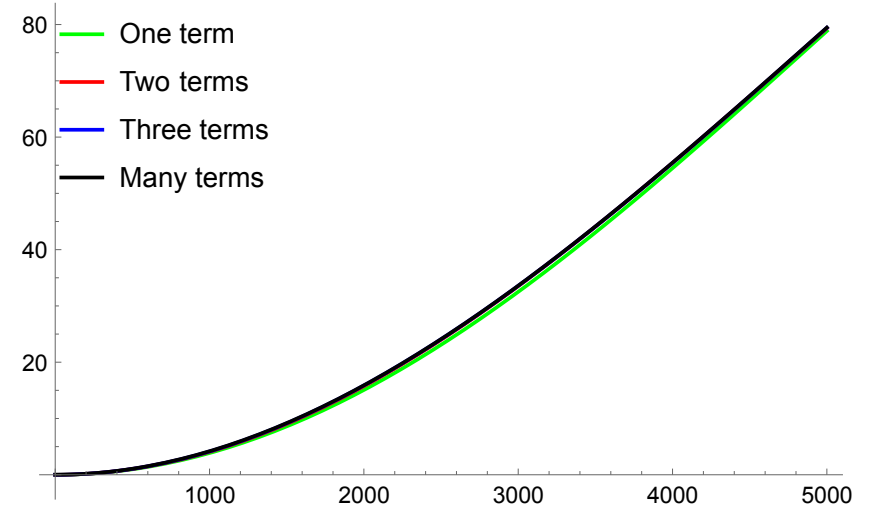
# Cantilever with Axial Force

See example at [terje.civil.ubc.ca](http://terje.civil.ubc.ca)

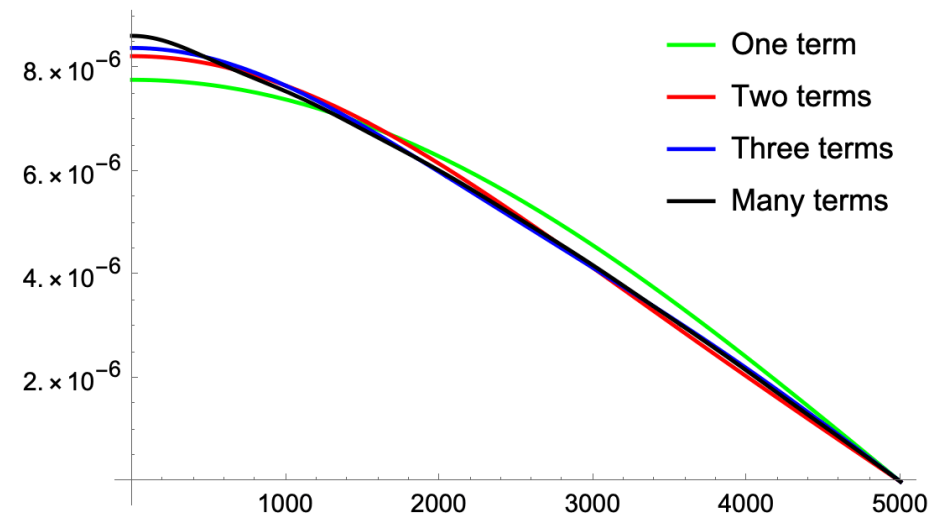


$$\text{Ritz: } w(x) = \sum a_n \cdot \left(1 - \cos\left(\frac{n \cdot \pi \cdot x}{2L}\right)\right)$$

Displacement,  $w(x)$ :



Curvature,  $w''(x)$ :



# Membrane Effect

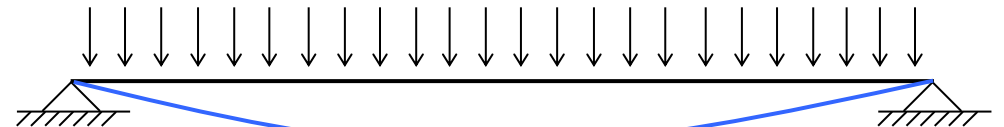
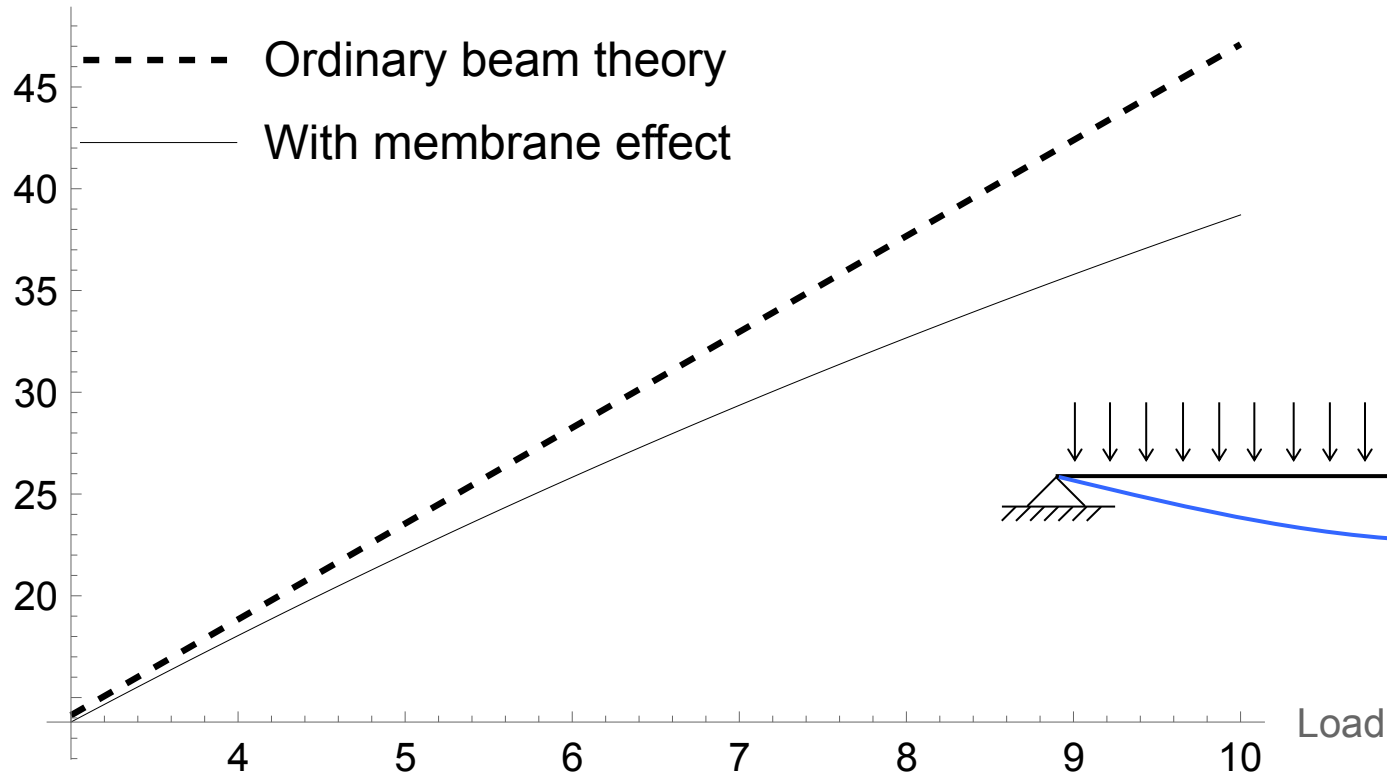
See example at [terje.civil.ubc.ca](http://terje.civil.ubc.ca)

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$$\varepsilon = z \cdot \frac{d^2w}{dx^2} + \frac{1}{2} \cdot \left(\frac{dw}{dx}\right)^2$$

$$H = -\int_0^L q_z \cdot w dx$$

Displacement



More lectures:

Terje's Toolbox:

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