# The Finite Element Method 

This video:<br>The Finite Element Method for Truss and Beam Elements

Terje's Toolbox is freely available at terje.civil.ubc.ca
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## Elements

## Boundary Value Problems



## Forms of the BVP



Weighted residual form


Weak form (virtual work)


Variational form (energy)

## Truss Member



## Forms of the Truss BVP

$$
\frac{d u}{d x} \equiv u^{\prime}
$$

Strong form (differential equation):

$$
E A \cdot u^{\prime \prime}+q_{x}=0
$$



Weighted residual form:

$$
\int_{0}^{L}\left(E A \cdot u^{\prime \prime}+q_{x}\right) \cdot \delta u d x=0
$$



Weak form (virtual work):

$$
\int_{0}^{L}\left(-E A \cdot u^{\prime} \cdot \delta u^{\prime}+q_{x} \cdot \delta u\right) d x=0
$$



Variational form (energy):

$$
\int_{0}^{L}\left(-\frac{1}{2} \cdot E A \cdot\left(u^{\prime}\right)^{2}+q_{x} \cdot u\right) d x=0
$$

## Weak Form = Virtual Work

Virtual work:

$$
\delta W_{\mathrm{int}}=\delta W_{\mathrm{ext}}
$$

Principle of virtual displacements:

$$
\int_{V} \sigma \cdot \delta \varepsilon d V=\int_{0}^{L} q_{x} \cdot \delta u d x
$$

$\sigma=E \cdot \varepsilon$ Substitute material law:

$$
\int_{V} E \cdot \varepsilon \cdot \delta \varepsilon d V=\int_{0}^{L} q_{x} \cdot \delta u d x
$$

$\varepsilon=\frac{d u}{d x}$ Substitute kinematic compatibility:

$$
\int_{V} E \cdot u^{\prime} \cdot \delta u^{\prime} d V=\int_{0}^{L} E A \cdot u^{\prime} \cdot \delta u^{\prime} d x=\int_{0}^{L} q_{x} \cdot \delta u d x
$$

Weak form from previous slide:

$$
\int_{0}^{L}\left(-E A \cdot u^{\prime} \cdot \delta u^{\prime}+q_{x} \cdot \delta u\right) d x=0
$$

## Discretization


$N_{1}(x)=1-\frac{x}{L}$
${ }^{N_{2}} N_{2}(x)=\frac{x}{L}$

$$
\underbrace{\left(\int_{0}^{L} E A \cdot \mathbf{N}^{{ }^{T}} \mathbf{N}^{\prime} d x\right)}_{\mathbf{K}} \mathbf{u}=\underbrace{\int_{0}^{L} q_{x} \mathbf{N}^{T} d x}_{\mathbf{F}} \quad \square \quad\left[\begin{array}{cc}
\frac{E A}{L} & -\frac{E A}{L} \\
-\frac{E A}{L} & \frac{E A}{L}
\end{array}\right]\left\{\begin{array}{c}
u_{1} \\
u_{2}
\end{array}\right\}=\left\{\begin{array}{c}
\frac{q_{x} L}{2} \\
\frac{q_{x} L}{2}
\end{array}\right\}
$$

$$
\begin{aligned}
& \int_{0}^{L}\left(-E A \cdot u^{\prime} \cdot \delta u^{\prime}+q_{x} \cdot \delta u\right) d x=0 \\
& u(x)=\mathbf{N u}=\left[\begin{array}{ll}
N_{1}(x) & N_{2}(x)
\end{array}\right]\left\{\begin{array}{l}
u_{1} \\
u_{2}
\end{array}\right\} \\
& -\int_{0}^{L} E A\left(\mathbf{N}^{\prime} \mathbf{u}\right)\left(\mathbf{N}^{\prime} \delta \mathbf{u}\right) d x+\int_{0}^{L} q_{x}(\mathbf{N} \delta \mathbf{u}) d x=0 \\
& \delta \mathbf{u}^{T}\left(-\left(\int_{0}^{L} E A \cdot \mathbf{N}^{T T} \mathbf{N}^{\prime} d x\right) \mathbf{u}+\int_{0}^{L} q_{x} \mathbf{N}^{T} d x\right)=0
\end{aligned}
$$

## Beam Bending



## Forms of the BVP

Strong form (differential equation):

$$
E I \cdot w^{\prime \prime \prime \prime}-q_{z}=0
$$

Weight and integrate $\downarrow \quad \uparrow$ Require point-wise fulfilment

Weighted residual form:

$$
\int_{0}^{L}\left(E I \cdot w^{\prime \prime \prime \prime}-q_{z}\right) \cdot \delta w d x=0
$$

Integration by parts $\downarrow$ Integration by parts
Weak form (virtual work):

$$
\int_{0}^{L}\left(E I \cdot w^{\prime \prime} \cdot \delta w^{\prime \prime}-q_{z} \cdot \delta w\right) d x=0
$$



$$
\int_{0}^{L}\left(\frac{1}{2} \cdot E I \cdot\left(w^{\prime \prime}\right)^{2}-q_{z} \cdot w\right) d x=0
$$

## Weak Form = Virtual Work

Virtual work: $\quad \delta W_{\mathrm{int}}=\delta W_{\mathrm{ext}}$

Principle of virtual displacements: $\quad \int_{V} \sigma \cdot \delta \varepsilon d x=\int_{0}^{L} q_{z} \cdot \delta w d x$
$\sigma=E \cdot \varepsilon$ Substitute material law:

$$
\int_{V} E \cdot \varepsilon \cdot \delta \varepsilon d x=\int_{0}^{L} q_{z} \cdot \delta w d x
$$

$\varepsilon=-\frac{d^{2} w}{d x^{2}} \cdot z$
Substitute kinematic compatibility:

$$
\int_{V} E \cdot w^{\prime \prime} \cdot \delta w^{\prime \prime} \cdot z^{2} d x=\int_{0}^{L} E I \cdot w^{\prime \prime} \cdot \delta w^{\prime \prime} d x=\int_{0}^{L} q_{z} \cdot \delta w d x
$$

Weak form from previous slide:

$$
\int_{0}^{L}\left(E I \cdot w^{\prime \prime} \cdot \delta w^{\prime \prime}-q_{z} \cdot \delta w\right) d x=0
$$

## Discretization



$$
\begin{aligned}
& \int_{0}^{L}\left(E I \cdot w^{\prime \prime} \cdot \delta w^{\prime \prime}-q_{z} \cdot \delta w\right) d x=0 \\
& w(x)=\mathbf{N u}=\left[\begin{array}{ll}
N_{1}(x) & N_{2}(x)
\end{array}\right]\left\{\begin{array}{c}
u_{1} \\
u_{2}
\end{array}\right\} \\
& \int_{0}^{L} E I \cdot\left(\mathbf{N}^{\prime \prime} \mathbf{u}\right) \cdot\left(\mathbf{N}^{\prime \prime} \delta \mathbf{u}\right) \mathrm{d} x-\int_{0}^{L} q_{z}(\mathbf{N} \delta \mathbf{u}) \mathrm{d} x=0 \\
& \delta \mathbf{u}\left(\left[\int_{0}^{L} E I \cdot \mathbf{N}^{\prime T} \mathbf{N}^{\prime \prime} \mathrm{d} x\right] \mathbf{u}-\int_{0}^{L} q_{z} \mathbf{N}^{T} \mathrm{~d} x\right)=0
\end{aligned}
$$

$$
\underbrace{\left[\int_{0}^{L} E I \cdot \mathbf{N}^{n T} \mathbf{N}^{n} d x\right]}_{\text {Sitfress matix, } \mathbf{K}} \mathbf{u}=\underbrace{\int_{0}^{L} q_{z} \mathbf{N} d x}_{\text {Load vector }, \mathbf{F}} \quad \square \quad\left[\begin{array}{cc}
\frac{4 E I}{L} & \frac{2 E I}{L} \\
\frac{2 E I}{L} & \frac{4 E I}{L}
\end{array}\right]\left\{\begin{array}{l}
u_{1} \\
u_{2}
\end{array}\right\}=\left\{\begin{array}{c}
-\frac{q_{z} L^{2}}{12} \\
\frac{q_{z} L^{2}}{12}
\end{array}\right\}
$$

More lectures:

Terje's Toobox:
terje.civil.ubc.ca

